

DECEMBER 5, 1921

# AVIATION AND AIRCRAFT JOURNAL

VOL. XI. NO. 23

*Member of the Audit Bureau of Circulations***INDEX TO CONTENTS**

Editorials .....	651	Admiral Sims on Aviation.....	659
Means for Improving Airplane Performance .....	652	"Who's Who in American Aeronautics".....	660
The French Anti-Aircraft Service .....	655	First Demonstration of the Alula Wing .....	662
Spain to Argentine Airship Service .....	656	French Aircraft Engine Competition.....	662
State Aeronautical Legislation .....	658	Aviation Policy of the American Legion.....	663
A Message from Gabriele D'Annunzio .....	658	Foreign Aeronautical News.....	664

THE GARDNER, MOFFAT COMPANY, Inc., *Publishers*

HIGHLAND, N. Y.

225 FOURTH AVENUE, NEW YORK

SUBSCRIPTION PRICE: FOUR DOLLARS PER YEAR. SINGLE COPIES FIFTEEN CENTS. CANADA, FIVE DOLLARS. FOREIGN, SIX DOLLARS A YEAR. COPYRIGHT 1921, BY THE GARDNER, MOFFAT COMPANY, INC.

ISSUED EVERY MONDAY. FORMS CLOSE TEN DAYS PREVIOUSLY. ENTERED AS SECOND-CLASS MATTER NOV. 22, 1920, AT THE POST OFFICE AT HIGHLAND, N. Y. UNDER ACT OF MARCH 3, 1897.

## THOMAS-MORSE AIRCRAFT CORPORATION



*Thomas-Morse Training 2-Seater  
in flight over Ithaca, N. Y.*

## THOMAS-MORSE AIRCRAFT CORPORATION



# Means for Improving Airplane Performance

By Harlan D. Fowler, A.E.

Ever since the first successful airplane took to the air, with its live load of one man and a few pounds of fuel, its designers and various ideas have been suggested to create more efficient means whereby the proportion of this live load may be increased and economically incorporated.

The only invention which was designed to decrease the weight of the engine and its fuel consumption. The engine which was used in the first Wright airplane in 1903 was constructed by the Wright brothers and developed 12 hp, and the weight was 125 lb. or 12.5 lb. per hp. By 1914 the average best engine gave 420 hp and weighed about 640 lb. or 2.2 lb. per hp. In the same period of time fuel consumption was reduced from 3.60 lb. per hp. hr. in 1903 to 0.90 lb. per hp. hr. in 1919. This great reduction in weight of aircraft engines and their fuel consumption has made possible the partial realization of the economical and efficient airplane, although it must be recalled that aeronautical engineering should be realized for the body developed airplane structure as it is known today. Further decrease in engine weight and fuel consumption is looked for but the fact has been clearly pointed out in recent years of the inevitable sacrifice of reliability and load capacity as the result of light material and modality in power units.

In the attempt to keep pace with the demand for better performance it was necessary to consider the possibility of the aerodynamic and reduction of resistance. The effort to which this reduction in resistance has been recently (as appears) in our present designs of various multi-engine wings and the streamlining of the fuselage and engine nacelles.

## Development in Aerobics

It is evident that to obtain any appreciable increase of airplane performance and its carrying ability that the aerodynamic forces the aerodynamic lift of airplane design.

The establishment of aerodynamic laboratories quickly brought about a surprising and far reaching discovery. It has always been assumed that the winging power of an aircraft was derived from the pressure of the air against its under surface, whereas it was found that the upper surface carried about 75 per cent of the load (Fig. 1), and was caused by the vacuum which existed there from the reflexion of the air from the leading edge of the wing. The lower surface is only subjected to pressure amounting to 25 per cent of the load. This proportion varies slightly for different angles and is individually determined separately. This discovery was led to the conception of much more efficient aerobics.

Wing curvature is a highly scientific development and is the determining factor in the performance of an airplane. An aircraft for load carrying is amenable for high speed, and vice versa. Therefore, we must compromise. It is essential that the landing speed of the airplane be sufficiently slow to permit safe landing, the landing speed the more expert pilot is required and the larger the landing field. This landing speed determines the maximum as well as the maximum speed. The lifting capacity of an aircraft is determined from its aerodynamic characteristics. The design range of the angle of incidence through which lift may be obtained is from minus two degrees to plus sixteen degrees in which the value of the lift coefficient ranges from zero to maximum. The lift coefficient and maximum angle of incidence determine the maximum speed.

## Combining High Speed with Large Load

Ordinarily, the landing speed is from 70 m.p.h. to 90 m.p.h., and the speed across about 70 m.p.h. Roughly this means that the maximum speed may vary from 110 m.p.h. for load carrying to 160 m.p.h. for speed across. Airplanes that attain a speed of 200 m.p.h. which has recently been attained must land at about 80 m.p.h. Speed reduction has poorer per load carried. Thus we have engines that weigh

about 500 lb. which develop 425 hp. This is a weight of 2 lb. per hp. This type of engine may be placed in an aircraft carrying a total of either 2100 lb. or 8000 lb., which will give a loading of 5 lb. or 20 lb. per hp., respectively. The same load carried per hp. the lower the maximum speed, because it is like a man carrying a heavy load which compels him to travel slow. It is thus clearly shown that if we desire high speed the load must be small, with high power and short wing area, the latter condition of which will compel a landing speed of a minimum nature. Now, then, can it not be possible to devise something that will permit our airplanes to carry the huge load and still obtain this high speed without impairing the safe landing speed? Present attempts seem to indicate that this can be accomplished and it is my endeavor in this article to show how we are trying to bring about this accomplishment.

## The Fundamental Formula

The fundamental formula of aerodynamics for sustentation is

$$L = C_L A V^2$$

where

$$L = \text{lift load of the airplane}$$

$$C_L = \text{lift coefficient, expressed in lb. per sq. ft. m.p.h.}$$

$$A = \text{total area of the wing}$$

$$V = \text{velocity of the airplane in m.p.h.}$$

and combined with adjusted power factor will mean

$$L = C_L A V^2$$

where  $L$  and  $V$  being constant values (it is evident that  $C_L$  and  $A$  offer the two avenues which may be varied to obtain a certain modulus of performance).

$A$  is the total structural element which can be altered and it was with this fact the first attempt of a mechanically controlled airplane wing was made.

The short law has been recognized for many years and various means were considered to alleviate this restriction. These may be divided as follows:

- (a) Variable area.
- (b) Variable camber.
- (c) Variable angle of incidence.
- (d) Aircraft design, incorporating positive contours and adjustable portions.
- (e) A combination of any one of these with the other.

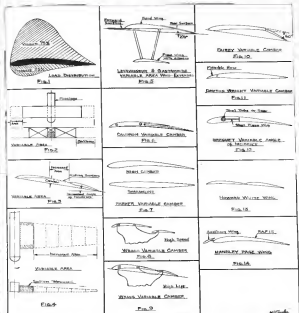
## Variable Area

(a) The methods of accomplishing this are numerous, but commonly consisted of extending the span of the wing as shown in Fig. 2. The disadvantage of this scheme was in order to obtain any appreciable increase in area the extension would be so great as to offer high stresses in the wing bones at the peak of the cantilever principle involved. The method of control was simple enough, while the increase in weight was small.

A patent was obtained recently in which two wings in tandem had a span left between them, this span 30 ft. fixed by a spring constant from the rear of the front wing. Fig. 3. In doing this it had the advantage of increasing the camber and the angle of incidence simultaneously, which is additional to increasing  $A$  tends also to increase the value of  $C_L$ , provided, of course, the contour of the topdell where fixed  $C_L$  is provided. However, the tandem principle has proved but little advantage so far, and for speed purposes it did not meet the requirements.

As the result of recent investigations in which wing a method was evolved in which the entire span was made to telescope, Fig. 4. This certainly offered an enormous possibility for increased area of the internal construction to take care of the cantilever stresses could be worked out.

The first practical demonstration of a variable area wing was made in the latter part of 1923 at Elampou, France. It was designed by Levassieur and Gustavich, two French



arrangement engineers. It was fitted to a tractor airplane with a 224 hp. Galloway engine. The novel design of this design enables the pilot to vary the surface, while in flight, from 340 sq. ft. to 950 sq. ft. or a ratio of 3 to 1. The area of the bottom wing is fixed, the upper wing being adjustable. The latter is composed of three parts, two of which are shifting. The middle section remains fixed, the lower one slides forward, while the upper one slides backward, Fig. 5. The third two parts from 6 ft. 5 in. to 10 ft. 6 in. The front panel, which is flat, displaced horizontally, while the rear portion which is curved, moves backward and forward. Thus the center, the wing and the angle of incidence. When fully open, the

leading edge is 12 in. below the leading edge, and the angle of incidence has passed from 3 deg. to 8 1/2 deg. The movement of the surfaces is made by means of cables. Its weight is 5,140 lb. fully loaded. Lift load (maximum) 3 lb. per sq. ft. (maximum) 0.5 lb. per sq. ft. Power loading 12.4 lb. per sq. ft. Maximum speed, 125 m.p.h., Landing speed, 37 m.p.h. Costing, 25,000 ft.

## Variable Camber

(b) The variable camber may next be considered. When properly proportioned, the camber of a wing is a large factor in increasing the value of the lift coefficient  $C_L$ .

Perhaps the most simple construction of this was suggested by the German design of France, which was considerably used by these two years previous to the war. The shape of their surface was flexible, the ribs of which are double reinforced for two-thirds of their chord from the leading edge, and only the outermost on the trailing edge, Fig. 6. The principle being that as the speed is increased the trailing portion would flap upward and change from a cambered to a high speed wing.

Considerable experiments were made on a variable camber aerofoil invented by H. P. Parker by the National Advisory Committee for Aeronautics. It consisted of a rib that could be made to alter the curvature of an aerofoil from a highly cambered to one of extremely low camber, Fig. 7. It had a maximum lift coefficient of 0.0638 and a minimum drag of 0.00035. It had evident hydrodynamic characteristics but the construction offered many difficulties.

A somewhat similar idea was used in a British design. Another method was suggested by C. A. Wigg in which two aerofoils were placed in tandem as a replace as shown in Fig. 8. The first speed aerofoil was the forward and being cambered and slightly above the smaller aerofoil. For landing the rear portion of the front aerofoil was deflected downward and closed in at the nose of the rear aerofoil, which was simultaneously raised, Fig. 9. The resultant camber of the aerofoil was decreased accordingly.

A very popular method of varying the camber is by means of raising a flap the whole span of the wing back of the rear spar, similar to the aileron, Fig. 10. This flap could be dropped 50 deg. and the lift coefficient of the main plane 10 deg. and obtain the maximum lift coefficient 0.06380. This wing is being used on the Fokker airplanes in Holland, with promising results, although it is said that they will be dropped 50 deg. for an emergency flap work in conjunction with the flap mechanism. The tests on the wind tunnel were carried on with an R.A.F. 9 aerofoil.

A more promising wing is one in which a narrow front flap and a fairly wide rear flap are attached to the aerofoil, both being moved simultaneously so that the new chord is always parallel to the original. Tests recently made at the NACA on a Giesse Martin 2-2P showed a maximum  $C_L$  of 0.0635.

Comparable attention was given to the Dornier Wright entry in the Gordon Bennett Aviation Cup race in September, 1909. The leading edge of the main plane was a flap principle, which had a limited ability of changing the aerofoil from a streamline to a cambered wing (Fig. 11). It met 80 coefficient is not known.

#### Variable Angle of Incidence

(a) The principle of a variable angle of incidence aerofoil was based upon the mechanical fact that the parasite resistance is approximately twice that of the lift when it is at the line of the wind. When the surface flaps at a higher angle and the leading edge is at a greater angle to the wind, the parasite resistance increases accordingly. In the variable angle of incidence machine, the flap is raised and lowered in the line of the wind with a theoretical gain in efficiency.

Perhaps the first practical attempt to utilize this principle was made by Ernest of France, and consisted of a method of raising a number of flaps at different angles to the line of the wind on different parts of the wing. This provided flexibility in the rib, as well as a mechanical method of reviving a single flap for the purpose of changing the angle of incidence. Tests with the Dornier machine resulted in a small gain.

Paul Schmitt, also of France, brought out an airplane in which the entire wing could be changed in angle of incidence. This machine, which was a biplane, was very popular in its day and had considerable success.

The Lemaire machine, constructed in this country, was very similar to the Schmitt type.

#### New Aerofoil Section

(40) Under this heading the field has been but little touched. At the French Olympia Show held in London in 1903 a biplane was exhibited by Howard Wright, the wing section of which may be described as made of two aerofoils, one of which in the leading edge of one plane is attached to the trailing

edge of the other. They were of such a shape, however, that the trailing edge of the lower surface of the upper regular curve, while the upper surface had a cambered dip at the center (Fig. 12). One of the reasons for the adoption of this machine was that the movement of the center of gravity was not affected by the change of incidence, 0 deg. and 10 deg. The machine was designed to fly in a flat air at an angle of incidence of -1 deg.

Perhaps the most recent idea of this has been suggested by that of P. Handley Page of England, and to have been been invented in Germany by Lachmann about the same time. As stated by Mr. Page, the principle utilizes the theory that the wing deflects the air through an angle so as to obtain the same force due to the change of momentum. The principle it has been found by experience that a thick or highly cambered wing has a larger lift per unit surface than a less cambered or flatter wing, although the highly cambered wing has a larger drag. Handley Page's design is to have the air at small angles and at small values of the lift in the air. It is also known that when a wing is deflected at an angle to the air through which it passes or which strikes against it, the same force is obtained by the change of momentum of the air at small angle to a certain deflected angle which varies between 12 deg. and 23 deg. After the critical angle is passed the value of the lift coefficient of the wing decreases rapidly. This fact is of great importance, and for certain portions of the upper surface of the wing. As a well known, the total reaction on an aerofoil is the sum of the positive pressure on the underside and the negative suction on the top surface. If this suction is small, the reaction is very small, and the machine will be unable to overcome the drag. The principle of this machine is to increase the lift by increasing the angle of incidence to angles greater than that mentioned, the maximum value of the useful lift coefficient will be increased. To obtain that, a small auxiliary flap, which is not attached to the main wing, but is a part of a slot arrangement can be made to extend forward (Fig. 13). In a public demonstration with this device recently attached to an H.I.V., a bank of very weak motion then another D.H.9, which took off at the same time. Its angle of attack was much steeper, though slower in speed on climb. In horizontal flight its speed was not improved, and on landing it seemed to maintain, but settled down gently and stopped within a few seconds. It is now reported that a new machine is being constructed incorporating this device. It seems that the more accurate wings there are the higher the lift obtained. For a single slot on an R.A.F. 15, the maximum lift coefficient is 0.06380, from 6 deg. to 10 deg. and the lift coefficient is 0.06380. The lift coefficient is higher with the slot open at all angles above 12 deg. Higher values have been obtained, but the aerofoil it was used with was not suitable for high speed flying.

#### Combining Various Features

(41) From the above review of various methods to obtain better efficiency it is clear that much attention has been devoted to this subject. It was not possible to include all of the known devices because of the length of the description, but it is hoped that the reader will find the subject of this article described. It is apparent, however, that any method which incorporates any of these ideas into a combination will make much more substantial progress, providing the mechanical construction can be easily worked out.

#### Comparative Analysis

Heretofore this review it is desired to know to what extent the various machines have been successful in their performance of the present day airplane. For most of their inventors it is difficult to make a comparative estimate, as little, if any, wind tunnel test data are available. However, the machine of P. Handley Page has been published, and the aerodynamic data of the latest design, and compared against the R.A.F. 22, the best overall efficient aerofoil as far developed. In Table I are given the aerodynamic characteristics, and the lift and drag coefficients of each of the machines, and the assumed airplane weighing 5000 lb. and capable of landing

December 5, 1931

at 20 m.p.h., also their value for high speed purposes, based on a maximum speed of 100 m.p.h.

In analyzing the relative efficiency, no allowance is made for the added weight of possible landing resistance which is inherent in the design of the device for controlling the lifting surface of the wing, for they are all new or less subject to the restriction. On the other hand, the ability of these systems to carry more weight than the tried ones is greatly proved from that of a highly cambered wing, and may lead to the nature statement of stress and strain, and thus present a much more efficient aerofoil, aerodynamically and structurally.

#### Conclusion

Of any of the devices so far produced it is evident that the variable area principle offers the most promising results. While no figures are available for the Lemaire and the variable camber sufficient information is available as to its landing speed, while the reaction of the wing when totally swept is available that of the R.A.F. 6, as it is mounted on the D.H.9, is available. It was 7.0. Some criticism may be directed to this value, but it will be found that area for an L/D of 41 corresponding to 150 lb. resistance and 40 m.p.h. that this figure of wing will offer a suitable improvement over any of the other devices considered. Another advantage of the form of variable area wing is the increased angle of incidence, in that device which means that the inclination of the chord is that same when landing, and consequently offering that much less resistance, and if there is sufficient strength of the popper blade the leading edge may be deflected, which is not possible with the other wing.

For landing, the Ailes Aerofoil must be the best with an L/D of 25.6, while for cruising speed the Handley Page is the best with an L/D of 16.0. However, this is so because of the form of the device, and the form of the wing. There is no doubt that this will be found to be a much better design. On climbing on account of the ability to utilize its large wing area, and resulting in carrying a much higher absolute ordinate. For cruising speed will be about normal, or about proportional to its high speed.

#### Some Minor Points

Several questions may be raised in regard to the strength and stability of a variable-area wing. From my knowledge

Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil

Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil
Continental Aerofoil	Handley Page Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil	Variable Area Aerofoil

## Spain to Argentine Airship Service

To Use Spanish Built Zeppelins of 6,300,000 cu. ft. Capacity

According to newspaper dispatches the long projected airship service between Spain and the Argentine Republic seems to have further advanced toward its realization. This scheme was suggested by Major Emilio Herrera, a well known Spanish balloonist and aviator, who suggested about a year ago to organize between Cadix and Buenos Aires a passenger and mail service by the use of semi-rigid airships of the Aeronaves type. The ships were to be built in Spain

has already subscribed 50 million pesetas, and that the Spanish government will guarantee the payment of 5 per cent dividends. The intervention of the Spanish government, it was stated, is made possible by a new law which authorizes the expenditure of public funds for the furtherance of ventures promising public utility.

Three Zeppelins are to be constructed at an approximate total cost of \$7,000,000 and about \$6,000,000 is to be spent



THE AIRSHIP STANDING AT STAVES, NEAR BERLIN, WITH THE COMMERCIAL ZEPPELIN "DIECKMANN" OVERHEAD, ABOUT TO LIFT

under the supervision of L. Torres Quevedo, the inventor of the type of construction, who is a Spaniard.

Nothing appears to have come of this project, but it is now announced on the authority of Hugo Reissner, director of the Zeppelin Air Navigation Co. ("Dietrich"), that a passenger and express service between Spain and the Argentine Republic is to be organized in the near future, using Zeppelin airships.

### Ships to be Built in Spain

As Germany ceased, under the terms of the peace treaty, exports and airships until she has made good the world destruction, after the Armistice, of some naval equipment, it is proposed to build the airships required for this service under the supervision of German engineers in Spain and Argentina. The operating crews are to be chosen from the staff of the Zeppelin works at Friedrichshafen, and the building materials are likewise to come from Germany.

For the construction of these ships, and their operation, a Spanish syndicate is now being formed with a capital of 50 million pesetas (about \$15,000,000 at current value). It is stated that a group of Spanish bankers and industrialists

on territorial facilities. The ships are to have a capacity of 6,300,000 cu. ft., an overall length of 525 ft., and a maximum diameter of 125 ft. They will develop a high speed of 40 m.p.h., and have accommodations for sixty passengers. Provisions are also to be made for carrying a maximum of 300,000 postal packages and letters.

The distance between Cadix and Buenos Aires is about 7,000 miles, and the trips are to be made in thirty hours. While the route made the service contemplated being too far as yet here announced, it seems likely that it will run by way of the Canary and Cape Verde Islands to Pernambuco, Brazil, and thence to Buenos Aires. A possible alternate route would run by way of Casablanca, Morocco, and Dakar, French West Africa. In either case the longest average trip will have a length of 2,000 miles, from Africa to South America.

### The European Terminal

The European terminal is to be situated between Cadix and Sevilla, where a large hangar will be constructed, and at Buenos Aires two sheds—one receiving—will be built. It is estimated that the construction of these sheds will take about two years, although the airships can be constructed in



THE TYPE OF PASSENGER ACCOMMODATIONS PROVIDED BY THE ZEPPELIN CO. FOR ITS LARGE COMMERCIAL AIRSHIPS. THE UPPER PART OF THE DRAWING IS A SPACIOUS, THE LOWER PART A LUGGER

much less time. Should the line run by way of Casablanca, a third stage would become available in the shape of a huge Zeppelin plant, owned by Germany, which the French Air Service proposes to rent at that place. The use of this air base would of course be conditional upon some form of French participation in the scheme, and while various hints have been made in some quarters to that effect, no definite announcement has as yet been forthcoming from France. In this connection it is likewise significant that the commercial airline service which is to be opened between Marseilles and Algiers, and which has the backing of the French Air Department, is to be run by the Compagnie des Messageries Transafricaines, a firm which is known to have connections with the Zeppelin company. It is therefore possible that this company may also be connected with the proposed South American

Zeppelin service, so that the latter would actually work out as a French-German-Spanish combine. The dearth of capital available in Spain, and the low rate of exchange of the German currency, because the likelihood of French capital being involved in the venture. If such should be the case, the South American strip the world probably be extended to Paris, where another of the co-owners, Zeppelin, stands in being. The use of the newly available air base at Orly, and perhaps even to Berlin. Regardless of possible commercial objections, France would derive considerably profit from such a service, the more so as there already exists a considerable passenger traffic between France and North America, while on the other hand it seems almost impossible that Germany could make such a venture without French participation of some sort or the other.

## Admiral Sims on Aviation

*Says Battleships Are Obsolete - Praises Airplanes and Aircraft Carriers*

The following remarks are taken from the Waterbury *Free Press* of Jan. 25 and report a talk given at Waterbury before the Second Congressional Church Forum.

That the battleship was becoming obsolete on account of the airplane, that the airplane carrier was becoming the backbone of the navy, and that scrapping of the battleships be built and planned, as proposed by the disarmament conference, would not affect the defenses of this country at all, were some

The great need now in the construction of airplane carriers, he said. The latest model had an absolutely new design, he said, the fuselage being on the sides of the ship. The deck was sufficiently wide and long to provide ample landing space. It had special elevators which carried the planes below decks. The wings of the planes could be folded up so that a large number of planes could be stored on several decks. An airplane carrier had nothing to fear from a battleship or

## State Aeronautical Legislation

In connection with the Watersbury Bill, which Representative Hahn introduced in the House, and the address by May W. J. Davis on the locality of Federal air legislation, the following summary of state air legislation, prepared by the Manufacturers Aircraft Association, is of timely interest.

The desirability of Federal air legislation which would provide for a uniform set of laws applicable to air navigation in all parts of the United States has repeatedly been pointed out in these columns. The large number of state air laws which are enumerated below show the urgent need of prompt Congressional action with respect to the Watersbury Bill.

In the enumeration which follows the first figure indicates the year in which the law was passed and the second (between parentheses) is the chapter or the law code of the state in question.

### California

1919 (1300). Prohibits the use of airplanes in landing.  
1921 (1343). An act concerning the registration, numbering and use of aircraft and the licensing of operators thereof. In effect Aug. 5, 1921.

### Connecticut

1918 (1318). Provision for the registration and regulation of aircraft. In effect June 8, 1919.

1920 (987). An act concerning aircraft, the register and rules hereon, in operation. Approved May 29, 1920.

### Indiana

1922 (1626). Provides penalty for the unlawful taking of aircraft. In force May 21, 1923.

1929 (107). Aviation field. Provides for acquisition, equipment, management, operation thereof.

### Kansas

1921 (204). An act creating a state aircraft board, regulating the navigation of the air by aircraft, establishing rules to provide maximum safety. Approved June 1, 1921.

1921 (164). Prohibits the use of airplanes in landing.

### Massachusetts

1919 (746). An act to regulate the use of aircraft. In effect 1923.

### Michigan

1919 (192). An act to prohibit the molesting of game and transgression kinds by operation of aircraft. April 15, 1919.

### Minnesota

1921 (475). An act to regulate the traffic of aircraft over lakes of the first class and prescribing penalties for the violation thereof. Approved April 29, 1921.

### Montana

1921 (335). Prohibits use of airplanes in landing.

### Nevada

1921 (165). An act to subordinate cities of the metropolitan class, cities, of 3rd class or over. 2nd class to acquire lands for the purpose of establishing an aviation field and to acquire the same and to provide the funds therefor by the sale of bonds of such city. Approved March 15, 1921.

### New Jersey

1921 (124). An act regulating the operation of motor-driven aerial machines. Approved March 31, 1921.

### New York

1919 (301-02). Insurance law now permits casualty, marine and fire insurance on "airplanes, dirigibles, or other aircraft."

1920 (735). An act to amend the greater New York charter, relating to zoning, granting and subjecting to the United States lands and lands under water, acquired by or owned by the city of New York, necessary for the establishment of air stations in connection with an defense of New York harbor and the Atlantic coast. May 11, 1920.

### New Carolina

1919 (38). Hearing held Govt with airplanes subjected.

### Oregon

1923 (133). An act to provide for appropriation of property for air field in value and laws. February 18, 1921.

1923 (141). An act to provide for the organization of an Oregon State Board of Aeronautics, regulating flying and registering system. Approved February 12, 1921.

### Texas

1919 (19). Authorizing private corporations for the purpose of construction, maintenance, etc., of airplanes. Approved February 17, 1919.

### Washington

1919 (148). An act relating to facilities for aerial transportation, authorizing cities and counties to acquire, maintain, and operate lands and property thereon, and declaring the same to be for public and not private and public use. Approved February 28, 1919.

### Wisconsin

1919 (313). Provides for readily landing fields.

### Wyoming

1917 (187). An act prohibiting the operations of airplanes, balloons, and other aircraft in the territory of Hawaii with certain exceptions. Approved April 15, 1917.

### A Message From Gabriele D'Annunzio

Gabriele D'Annunzio, Italy's Post-Aviation, who became a figure in international politics as the result of his command of Fiume, has given through Maj. James E. Chaney, Air Service, American Military Attaché at Rome, a message to the United States, a translation of which is as follows:

"While the wings of war birds are moving over the international lake for the Cup of the Eagles offered by me, I send across the corn air privileges and assignments of all Italian aviators to our glorious American compatriots who wish to see steering the aerial progress towards the promise of the future."

James E. Chaney, Major, D'Annunzio on Sept. 25 at Lake Garda, in which day were staged the international races for the D'Annunzio Cup.



"THE GREAT NEED NOW IN THE CONSTRUCTION OF AIRCRAFT CARRIERS"—ADMIRAL SIMS

of the startling statements made by Rear-Admiral W. S. Sims at the Second Congressional Church Forum.

The admiral referred indirectly to the controversy created by recent statements when he said that he proposed to speak on the question of battleships versus airplanes, aerostatics, etc., "as you see," he said, "there are explosive subjects, and still all possible chances for 'indiscretions.'"

"It has been said in the past that the battleship was the backbone of the fleet," he said, "but I believe it is no more. The battleship has no defense against an airplane but small anti-aircraft guns. At the present time it was considered that one bit in 1,000 was a good enough, with muzzled batteries for sides front at a place, and plenty of observation to determine the route. The best experts now agree that the results of anti-aircraft fire from a ship are negligible."

### Command of Air

"The command of the air means the command of the surface, whether it be sea or land. The experiment of making the battleship by airplane battles proves that of our coast is maintained by airplanes to ships can reach our shores or land troops."



TWO VIEWS OF THE FERRARIS ("PENTHOS"), A SHIPBOARD AIRPLANE OF THE BRITISH NAVY. THE FERRARIS IS MAJOR W. S. SIMS, AIR OF THE COMBAT, IN PARALLEL STRIKE ON BRIDGE DECK. THE LANDING DECK IS FILLED WITH BRITISH FLYING.

SAVING LIFE.  
Black Diamond

battle-planes, he said, as it had a speed of 35 knots—more than any battleship or cruiser—and could always loop out of the latter's range, while the airplanes if turned could destroy the battleships.

### "Battleship Gun"

"There are those of us who believe," he said, "that the battleship is gone and that scrapping them down is not only wise but necessary. We need not worry, either, about destroying the battle cruisers now building, as they are no help to the battle-ships against airplanes."

Aviators could be both far more rapidly and far more cheaply than ships, he said. If we spent millions on battleships and battle cruisers, and another million spent only half as much on airplanes, he said he would have no doubt of the result.

Numerous questions were asked at the conclusion of the address. One was:

"Will the battleship become obsolete because of airplanes?"

"Yes," Admiral Sims answered.







## Foreign Aeronautical News

## China

An official Chinese circular has been issued regarding the Peking-Tientsin airmail service which was inaugurated on July 1 and not Aug. 1, as was officially stated previously. Regulations governing the carriage of mail or parcels are briefly as follows:

- 1 The Peking-Tientsin airmail postal service will carry mail and valables. No passengers will be carried for the time being.
- 2 The service from Peking to Tientsin will be on Wednesdays, Fridays and Sundays, while from Tientsin to Peking the service will be on Tuesdays, Thursdays and Saturdays.
- 3 Mailmen will leave Peking on the scheduled dates at 5 p. m. while they will start on their return trip at 50.30 a. m. The journey will be covered in two and a half hours.
- 4 The Peking terminal station will be Nanyang temporarily, and that at Tientsin will be near Tientsin Tiao.
- 5 Besides the regular postage, mail or parcels carried by the service will pay a special postage, the amount of which will be announced by the Chinese Post Office.
- 6 Aerial stamps will be on sale at all Post Offices.
- 7 Ordinary mail and parcels will be received at all Post Offices, but valables must be taken for transportation in the preparation bag of the Peking-Tientsin airmail service administration, Peking, or either the Peking or Tientsin Aerodrome.

## Denmark

Denmark is engaged in organizing a system of international regulation air traffic. An agreement with Norway was recently signed, and a conference between Denmark and Germany delegates will begin at Copenhagen for the conclusion of a convention similar to that between Great Britain and Denmark. These arrangements will prove useful when aerial communication, which is at present nonexistent, comes to be re-established, as it will then be no longer necessary for aircraft to obtain a special permit to land on every occasion, as is now the case. Conventions on the same lines with Sweden and Finland are also being prepared.

## Egypt

An air mail service between Cairo and Baghdad has recently been inaugurated by the Royal Aerial Mail Force on the Middle East. This service will carry fortnightly, R.A.F. pilots and mailmen being employed to carry official mail. The scheme is arranged as part of the regular training of the R.A.F. The whole length of the line is 548 miles, the route from Cairo being via Rashidi, Assuan and Bahariya to Baghdad. A saving of ten to fourteen days will normally be effected by this service.

## Holland

The airmail service between Amsterdam and German cities has been discontinued. It will not resume. The service between Amsterdam and Paris and Amsterdam and London, which last year was discontinued from autumn to spring, will be continued without interruption through the winter. Apparently the service between Amsterdam and London, which is operated by a Dutch Company, is not economically profitable, as the Netherlands government grants it a considerable annual subsidy toward making up its deficit. The government estimates for the year 1932 make the amount of this subsidy fl. 370,000 (£145,748 at the normal rate of exchange).

## Sweden

Reports from Sweden indicate that it is intended to build an Aero House at Göteborg, Sweden, in 1932. The new international flight for the F-100 is July 1-30, 1932. It is stated that the Society of British Aircraft Constructors and the Chaudron Syndicate des Industries Aeronautiques (France) have been approached on the subject of participation in the Robinsons Ltd. task given Government orders. It is also hoped that Italy, Germany, Czechoslovakia and the United States will participate.

## Where to Fly

## CALIFORNIA

SAN FRANCISCO, CALIFORNIA  
EARL P. COOPER AIRPLANE & MOTOR CO.

## ILLINOIS

CHECKERBOARD AIRPLANE SERVICE  
FOREST PARK, ILLINOIS

## INDIANA

One of the largest and best equipped flying fields in the United States.  
CURTISS-INDIANA COMPANY  
Ellettsville, Indiana  
ALL TYPES OF CURTISS PLANE

## MASSACHUSETTS

BOSTON AND SPRINGFIELD, MASS.  
EASTERN AIRCRAFT CORP.  
245 FIRST ST., BOSTON, MASS.

## MINNESOTA

WHITE BEAR LAKE, MINN.  
The Twin Cities' first airmail route  
Harold G. Peterson Aircraft Company  
SCHOOL OF AVIATION

## NEW JERSEY

NEW YORK, AIR TERMINAL  
Two miles - 15 miles from New York  
Learns on ships that connect with you. Planes round 140 ft.  
CHAMBERLIN AIRCRAFT  
Immersion Day - July 1932 - N. Y. City

## NEW YORK &amp; NEW JERSEY

CURTIS FIELD, GARDEN CITY, LONG ISLAND  
KINGSTON FIELD, BUFFALO, N. Y.  
FLYING STATION, ATLANTIC CITY, N. J.  
CURTISS AIRCRAFT & MOTOR CORPORATION

## NEW YORK

AKRONAIRLINE AIRWAYS, INC.  
From Buffalo, New York  
To Pittsburgh, New York  
To Philadelphia, New York  
To Baltimore, Maryland  
To Washington, D. C.

## OHIO

Dayton, Cincinnati, Toledo and Columbus - 150 miles from Dayton, Ohio.  
JERSON AIRPLANE & SUPPLY CO.

## OREGON

LAND OF WATER FLYING  
ORIGEN, WASHINGTON AND OREGON AIRPLANE COMPANY  
PORTLAND, OREGON

## PHILADELPHIA

Flight School and Commercial Aviation  
and for Civilian  
Official Flying Field Area, City of Philadelphia  
PHILADELPHIA AERO-SERVICE CORPORATION  
608 East State Street Building, Philadelphia.

## PITTSBURGH

CLINTON WISCONSIN AIRCRAFT CO.  
FLYING SCHOOL  
Midwestern Air Port  
CHIEF E. M. JOHNSON

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.

220 CHERRY STREET, PITTSBURGH, PA.



## Submerged, four months—

FLYING down from New York to Miami last December, the Aero-Limited Company's hydroplane "Comet" developed motor trouble off Chincoteague, Va. The motor was removed and sent to Miami—the plane left anchored in Chincoteague Harbor.

It was not until May 18th that the Company learned the "Comet" had sunk sometime in January, and save for the upper wing, had remained submerged since that time.

The ship was raised and an unexpected leak in a vent tube stopped. As soon as the new motor was installed, the plane was flown to Flushing Bay, Long Island, without further repairs or even re-varnishing. For careful inspection showed that despite the four months' submergence in salt water, the wings were still absolutely fast, the hull in perfect condition—the varnish unaffected.

Amazing? Not when you remember that "Comet," like all "Aero-Limited" planes, was

*Valsparred, of course!*

## VALENTINE &amp; COMPANY

Largest Manufacturers of High-Grade Paints in the World—East, West  
New York, Chicago, Boston, Toronto, London, Paris, Amsterdam  
W. F. POLLOCK & CO., Portland, Maine



25 Centimeter Insertions \$20.00

## "The Bombs Heard 'Round the World"

When, to the amazement of the majority of observers on the transport Henderson, the ex-German battleship *Ostfriesland* sank in fourteen minutes from the effect of two 2,000 pound Army aircraft bombs, the doom of navies was not sealed, but aircraft as a third arm of national defense received the world-wide recognition which until then had been denied. This fact will have tremendous influence upon development of the aircraft industry.

IT WILL PAY YOU TO KEEP CLOSELY IN TOUCH WITH THESE DEVELOPMENTS AND COMPLETE TECHNICAL DETAILS BY READING EACH WEEK THE RECOGNIZED AMERICAN AUTHORITY ON AERONAUTICS—

Sign the coupon now but do not send any money with it. We will send you the next two issues as they appear and bill you for a year's or six month's subscription, as you prefer. If, upon receipt of bill, you decide not to continue, advise us promptly and accept the two issues with our compliments.

## Aviation and Aircraft Journal

GARDNER, MOFFAT CO., INC.,  
225 Fourth Ave., New York.

Send me the next two weekly issues of AVIATION AND AIRCRAFT JOURNAL. If, upon receipt of bill for subscription as indicated below, I decide not to continue, I will so advise you, and understand that no charge will be made for the two copies.

One Year (52 issues—U. S. \$4.00; Canada \$5.00; Foreign \$6.00)  
Six Mos. (26 issues—U. S. \$2.00; Canada \$2.50; Foreign \$3.00)

Name

Address



## The Laird Swallow

*America's First Commercial Airplane*

THE Swallow's carrying capacity—three passengers, baggage and full fuel load—is one of the reasons that it is fast becoming America's most popular commercial ship. Speedy, dependable and economical. Let us send you our booklet.



-- Now Priced at \$4,500 --

E. M. LAIRD COMPANY

MANUFACTURERS  
WICHITA, KANSAS  
General Sales Office  
2216 SO. MICHIGAN AVE., CHICAGO.

E. M. Laird Co.  
2216 So. Michigan Ave., Chicago, Ill.  
Please send me details of your new  
airline plane and copy of booklet.  
Name   
Address   
City

## EDWARD P. WARNER

Consulting Aeronautical Engineer

Mass. Institute of Technology  
Cambridge, Mass.

The Spark Plug That Cleans Itself

## B. G.

Contributed by the U. S. Army Air Service & the U. S. Navy

THE B. G. CORPORATION  
33 GOLD STREET  
NEW YORK CITY U. S. A.

## FOR SALE

Cellulose Acetate Dope, quality guaranteed, 50 gal. bbls. \$1.00 gal. gal. lots \$2.00 gal.

MAX TOPPER & ROSENTHAL  
E. 11th AVE. & P. R. TRACKS  
COLUMBUS, OHIO

## SAFETY FIRST AIRPLANE CLOTH

Pinked Tape — All Widths

TESTED AND GUARANTEED  
To Grade A Specification 16,004

*Immediate Delivery*

**W. HARRIS THURSTON & CO., INC.**  
116 Franklin St. New York

Telephone Franklin 1234  
*Contractors to U. S. Army and Navy*

## Flottorp Manufacturing Co. AIRCRAFT PROPELLERS

Established 1912



213 Lyon St., Grand Rapids, Michigan

*Contractors to United States Government*

# HAMILTON PROPELLERS

**HAMILTON AERO MFG. CO.**  
**MILWAUKEE, WIS.**



**ROEBLING**

Aircraft Wire, Strand and Cord

Thimbles and Ferrules

**JOHN A. ROEBLING'S SONS CO.**  
**TRENTON, N. J.**

## *Just A Few More Left* **Aeronautical Engineering and Airplane Design**

By LIEUTENANT ALEXANDER KLEMIN

Air Service, Aircraft Production, U. S. A., in Charge Aeronautical Research Department, Airplane Engineering Department. Until entering military service in the Department of Aeronautics, Massachusetts Institute of Technology, and Technical Editor of AVIATION and AERONAUTICAL ENGINEERING. In two parts.

### Part 1. Aerodynamical Theory and Data

Modern Aerodynamical Laboratories  
Elements of Aerodynamical Theory  
Sustention and Resistance of Wing Surfaces  
Comparison of Standard Wing Sections  
Variations in Profile and Plan Form of Wing Sections  
Study of Pressure Distribution  
Biplane Combinations  
Triplane Combinations—Uses of Negative Tail Surfaces  
Resistance of Various Airplane Parts  
Resistance and Comparative Merits of Airplane Struts  
Resistance and Performance  
Resistance Computations—Preliminary Wing Selections

### Part 2. Airplane Design

Classification of Main Data for Modern Airplanes; Unarmed Land Reconnaissance Machines; Land Training Machines  
Land Pursuit Machine; Land Gun-Carrying Machine; Twin-Engine All-round Machine  
Estimate of Weight Distribution  
Engine and Radiator Data  
Materials in Airplane Construction  
Worst Dynamic Loads; Factors of Safety  
Preliminary Design of Secondary Training Machine  
General Principles of Chassis Design  
Type Sketches of Secondary Training Machine—General Principles of Body Design  
Wing Structure Analysis for Biplanes  
Notes on Aerial Propellers

Price, Postpaid, in the United States, \$5.00 Net

**THE GARDNER, MOFFAT COMPANY, Inc., Publishers**  
**225 Fourth Ave., New York City**